

Code No: C7607

**JAWAHARLAL NEHRU TECHNOLOGICAL UNIVERSITY HYDERABAD**  
**M.TECH I SEMESTER EXAMINATIONS APRIL/MAY-2012**  
**FLIGHT VEHICLE STRUCTURES**  
**(AEROSPACE ENGINEERING)**

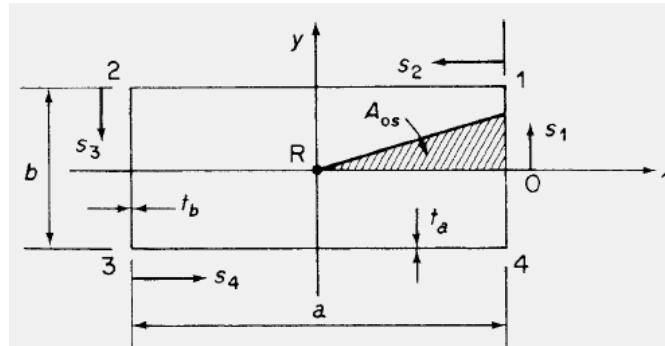
Time: 3 hours

Max.Marks:60

**Answer any five questions**  
**All questions carry equal marks**

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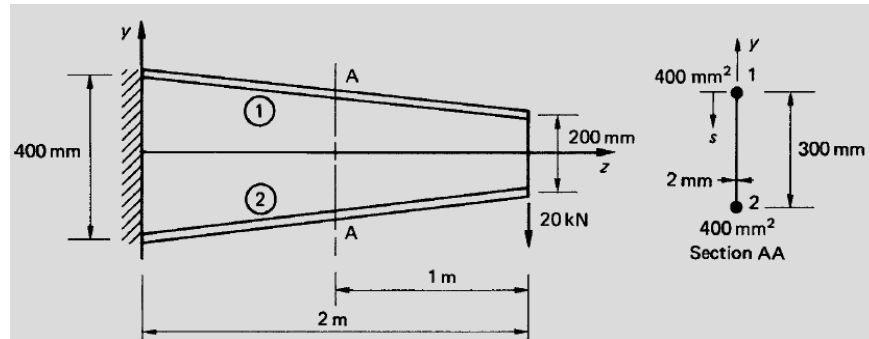
1. Explain the following with the help of figures.
  - a) Loads on aircraft structural components.
  - b) Functions of aircraft structural components.
  - c) Pressure distribution over an airfoil.
  
2. Determine the warping distribution in the doubly symmetrical rectangular, closed section beam, shown in figure below, when subjected to an anti-clock wise torque T.



3. Derive the equations to find out the shear forces per unit length of a plate subjected to a distributed transverse load.
  
4. Derive an equation for the critical stress in a thin flat plate due to buckling.
  
5. Determine the neutral axis position and direct stress of a thin walled beam due to bending.
  
6. Explain shear center. Derive the equation to find out the shear of open section beams.
  
7. Determine the shear flow distribution in the web of the tapered beam shown in figure below, at a section midway its length. The web of the beam has a thickness of 2 mm and is fully effective in resisting direct stress. The beam tapers symmetrically about its horizontal centroidal axis, and the cross-sectional area of each flange is  $400 \text{ mm}^2$ .

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- 8.a) Explain structural health monitoring.
- b) Explain different applications of smart materials.

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